

# Bi-Sat Observations of the Lunar Atmosphere above Swirls (BOLAS):

## Tethered SmallSat Investigation of Hydration and Space Weathering Processes at the Moon

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Planetary Science Deep Space  
SmallSat Mission Concepts



# BOLAS Science Targets and Rationale

The overarching science goal of the BOLAS mission is to determine the role of the **solar wind** in **space weathering** and the creation of **water products** on the surface of the Moon by investigating **crustal magnetic fields** and the **swirl patterns** that typically accompany them.

## Primary science:

- **Space weathering** – regolith exposure to solar wind protons
- **Lunar water cycle** – role of solar wind in forming OH/water
- **Solar wind interaction with crustal fields**

## Secondary science:

- **Crustal magnetism**
- **Effect of impacts on regional magnetism**
- **Moon's global interaction with the solar wind**
- **Exospheric dust transport**
- **Lunar ionosphere**

## Primary target:

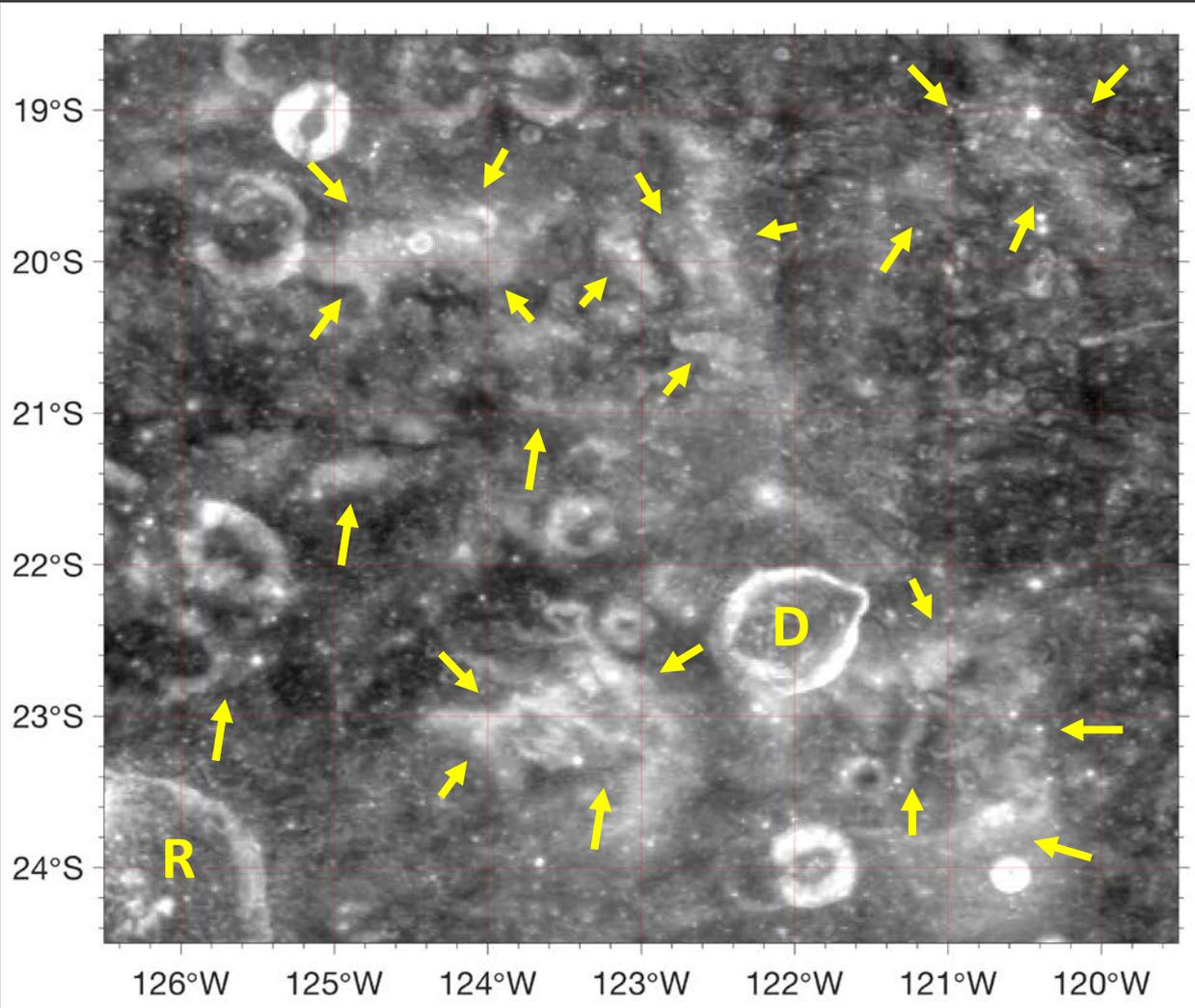
**Gerasimovich** – amongst strongest crustal magnetic fields with extensive swirl patterns. Located on farside around 21°S, 123.5°W

# BOLAS Science – Swirls, Space Weathering and Water

WAC Reflectance (643 nm)

Gerasimovich

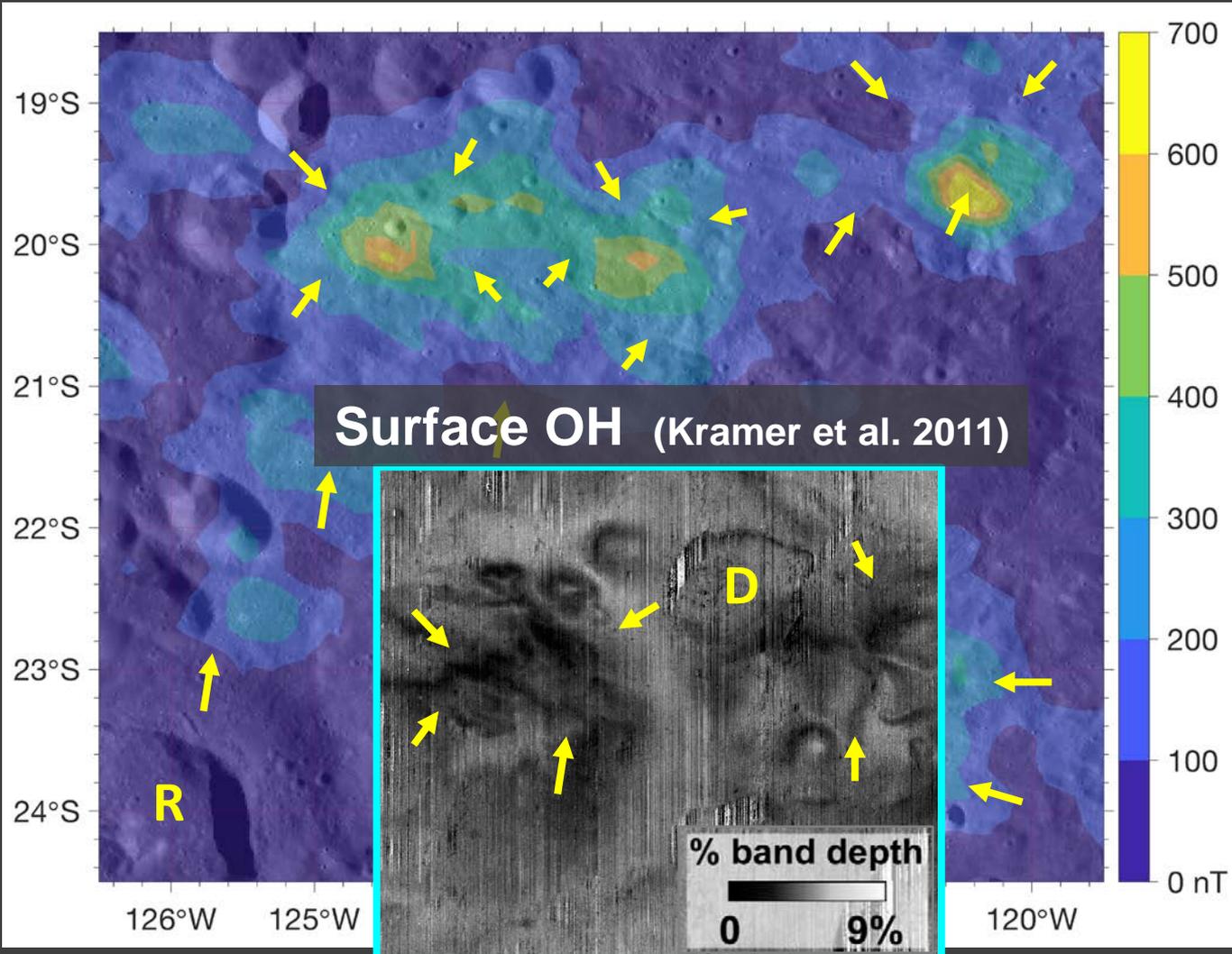
visibly bright  
“pristine” swirl  
patterns



# BOLAS Science – Swirls, Space Weathering and Water

Surface Magnetic Field [ nT ] (Tsunakawa et al. 2015)

Gerasimovich



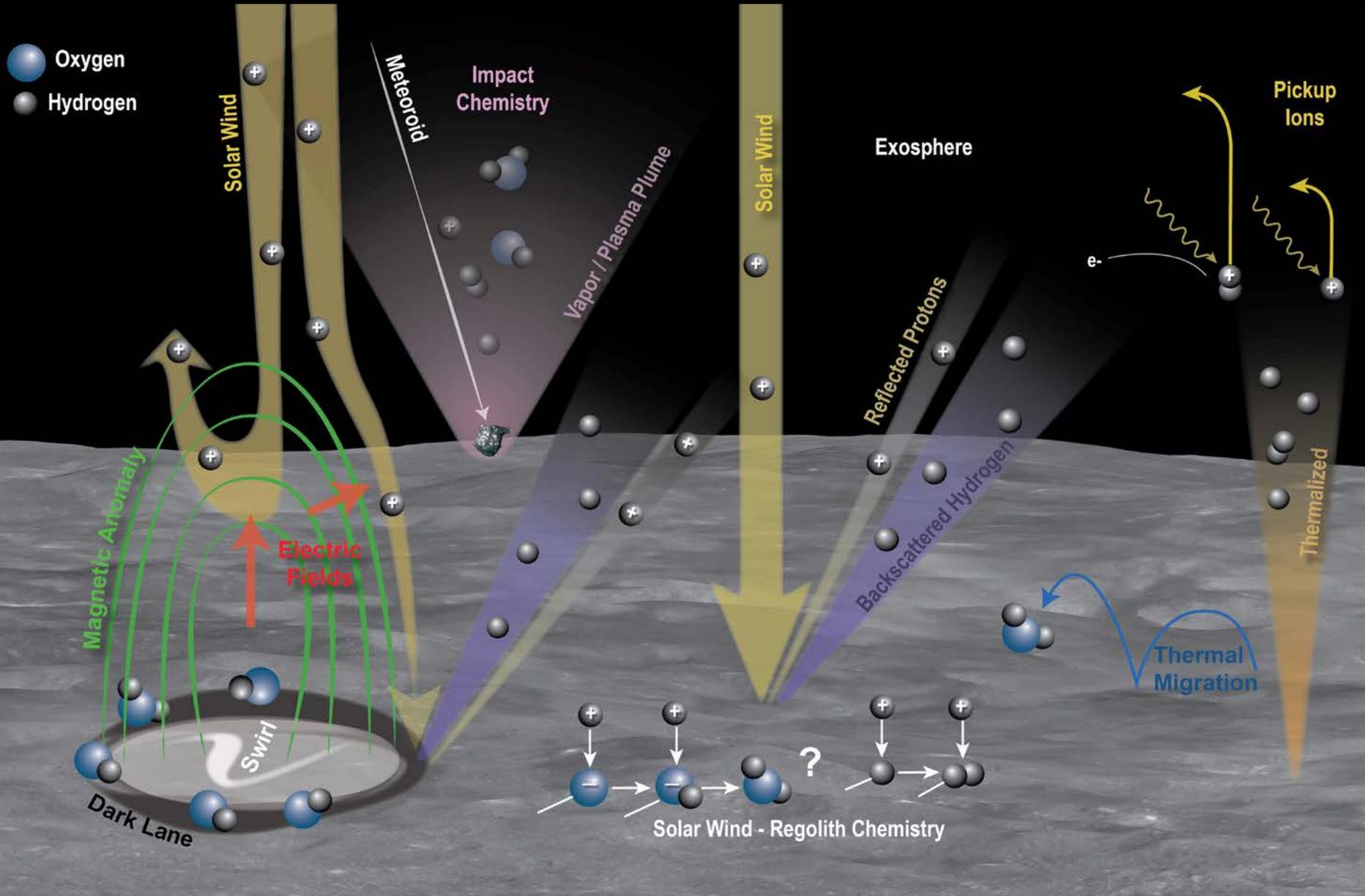
visibly bright  
“pristine” swirl  
patterns

coincide with  
strong crustal  
magnetic fields

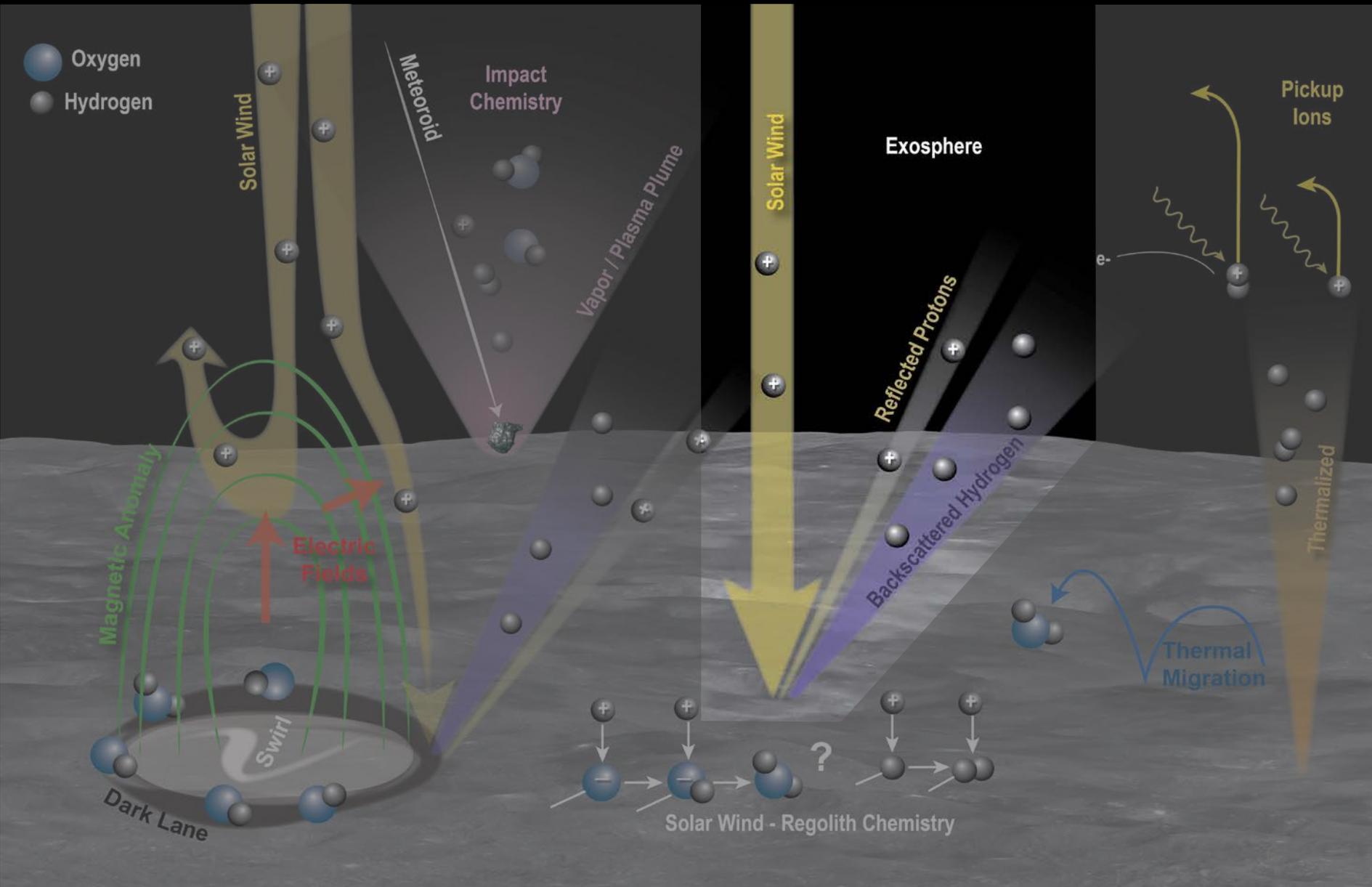
coincide with  
absence of OH

indicates that  
crustal fields  
shield surface  
regolith from  
space weathering  
and OH formation  
by the solar wind

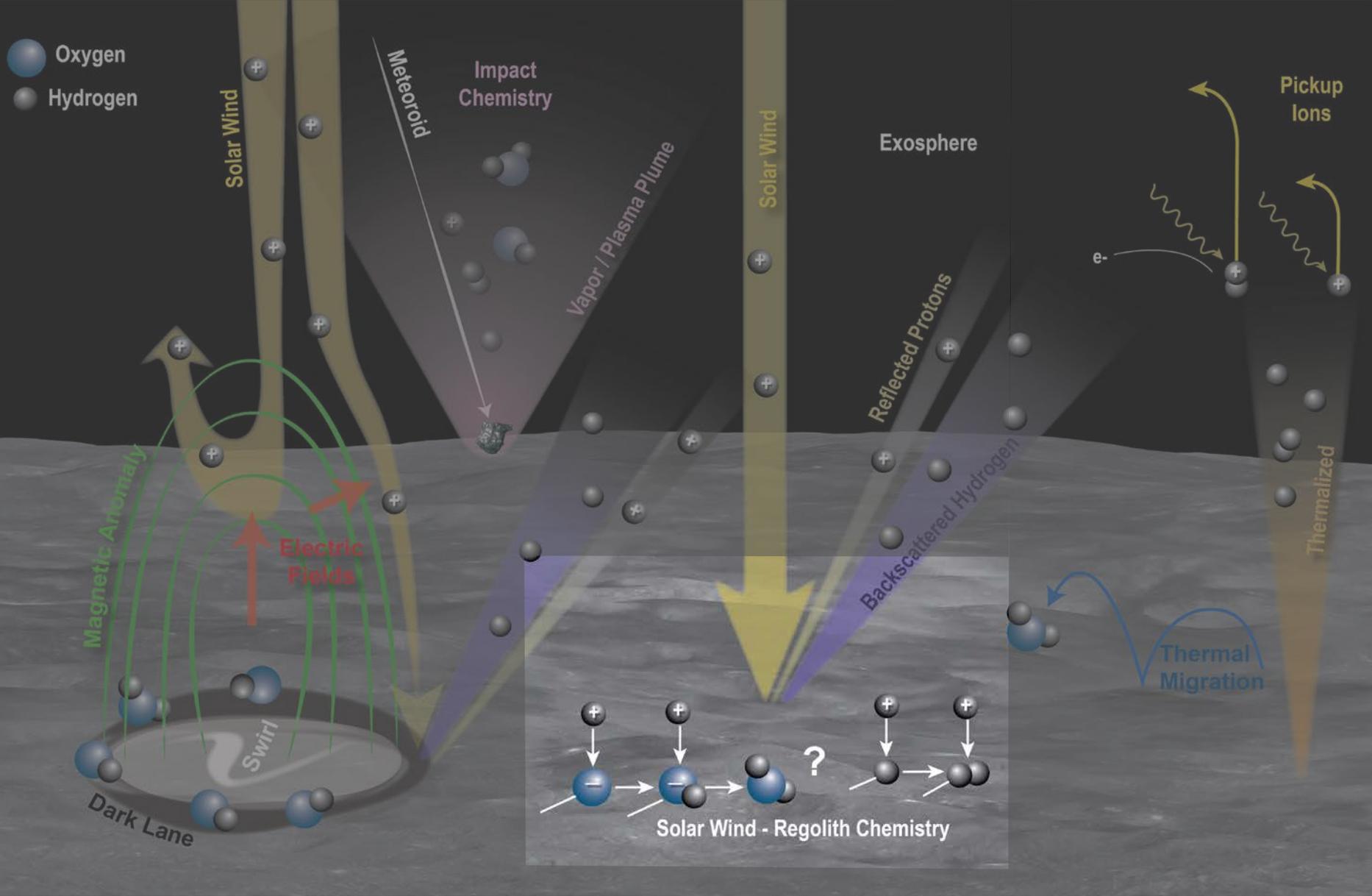
# BOLAS Science – Physical and Chemical Processes



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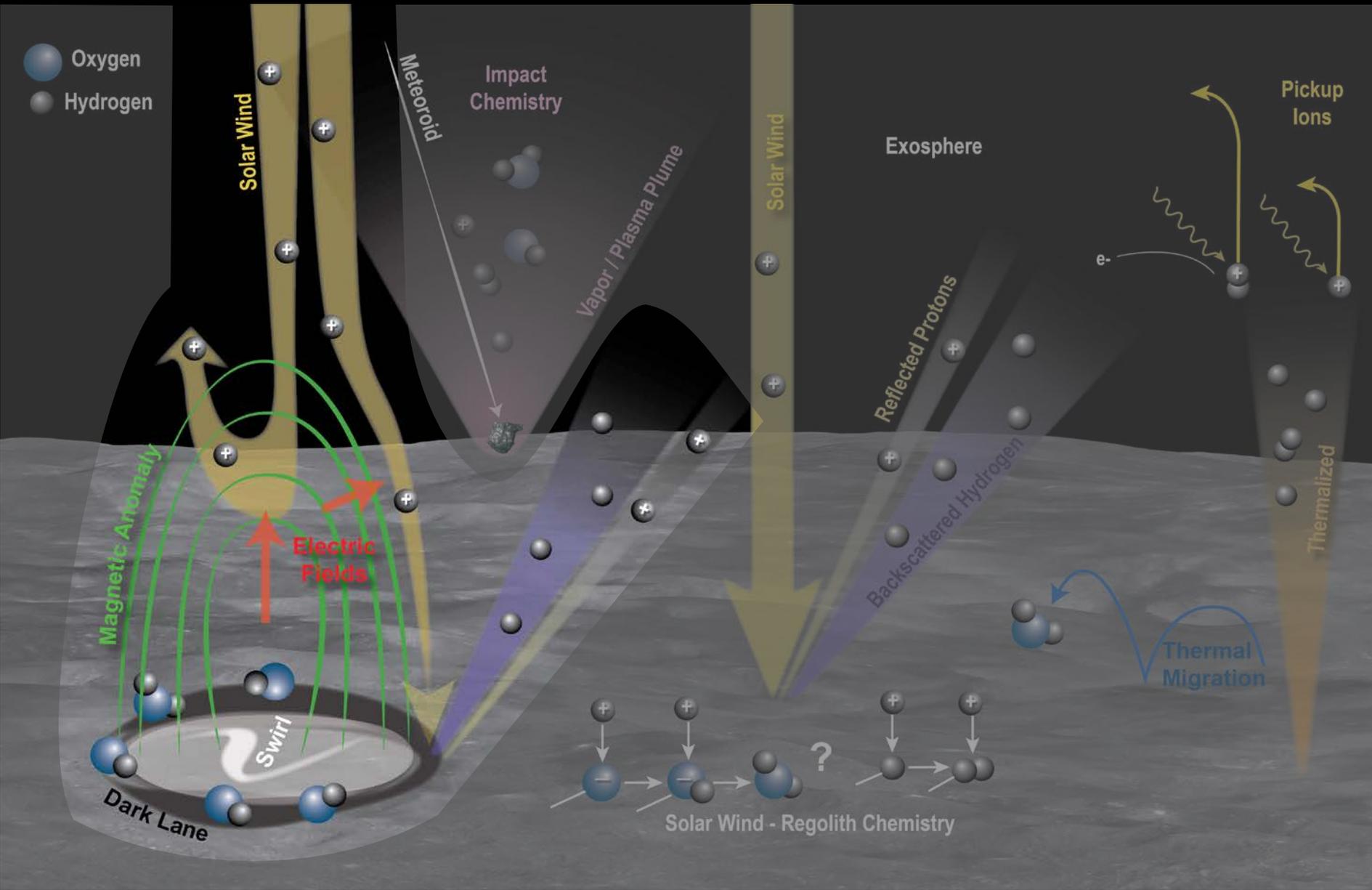


# BOLAS Science – Physical and Chemical Processes

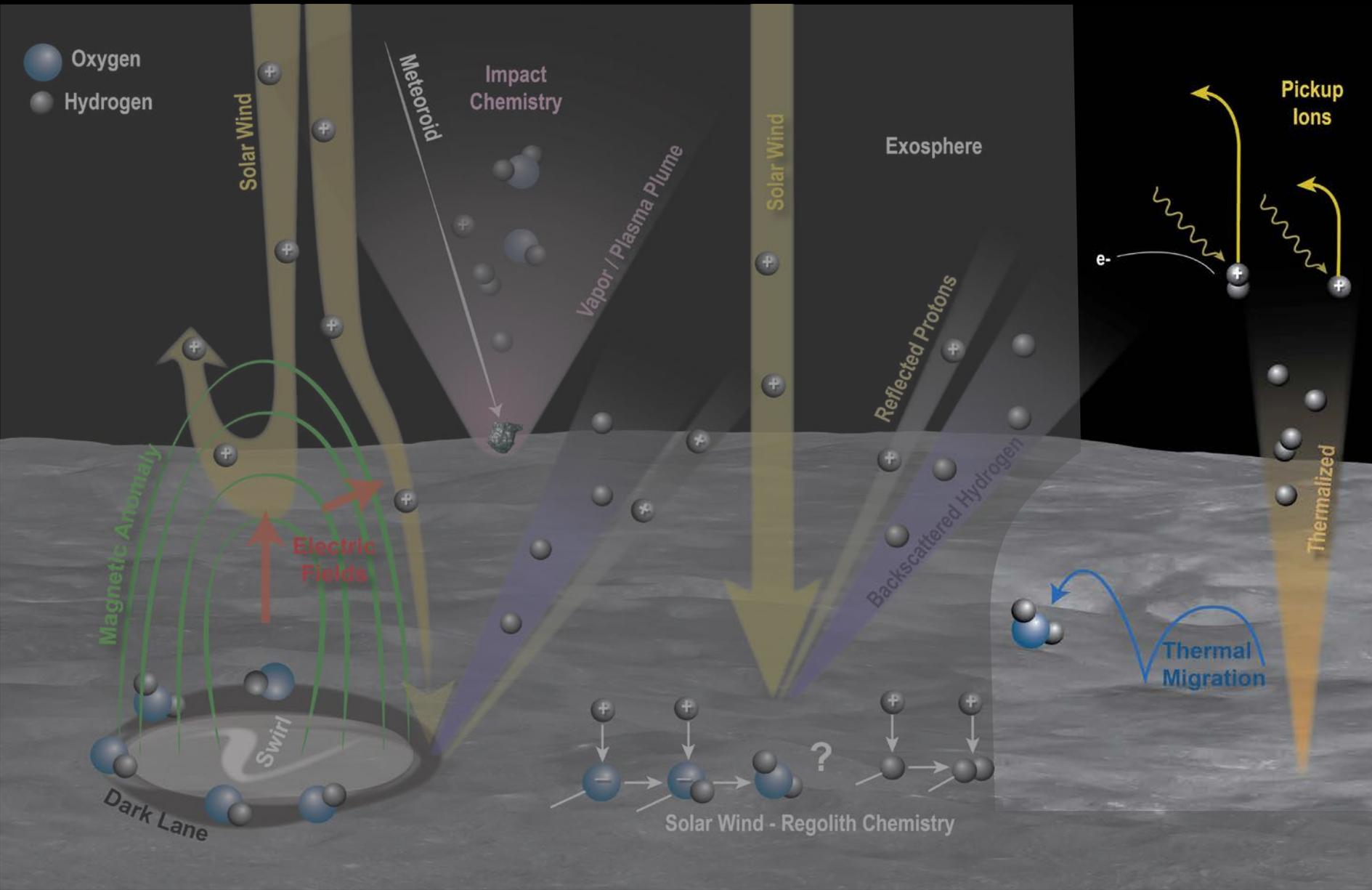


# BOLAS Science – Physical and Chemical Processes

- Oxygen
- Hydrogen

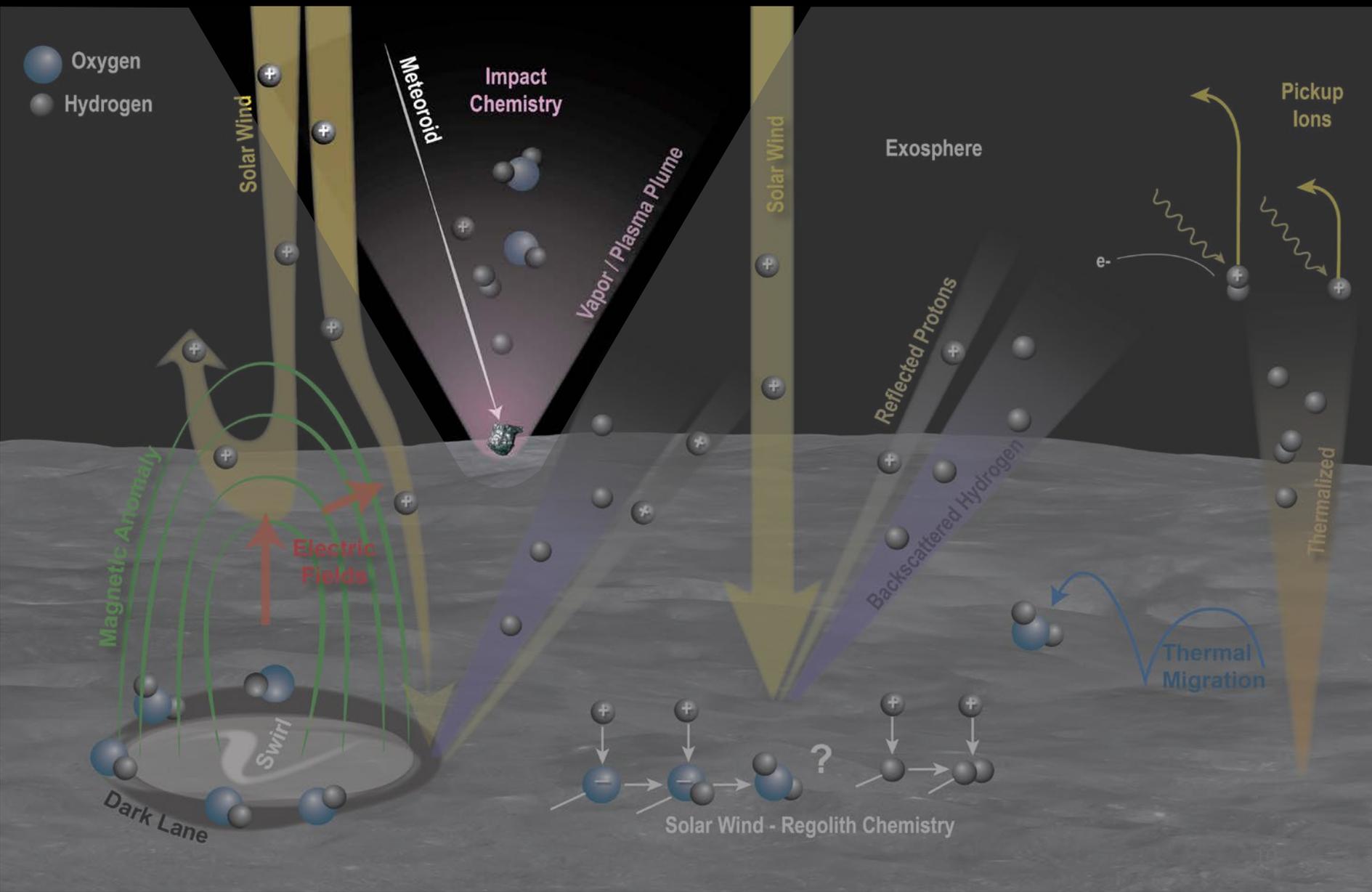


# BOLAS Science – Physical and Chemical Processes



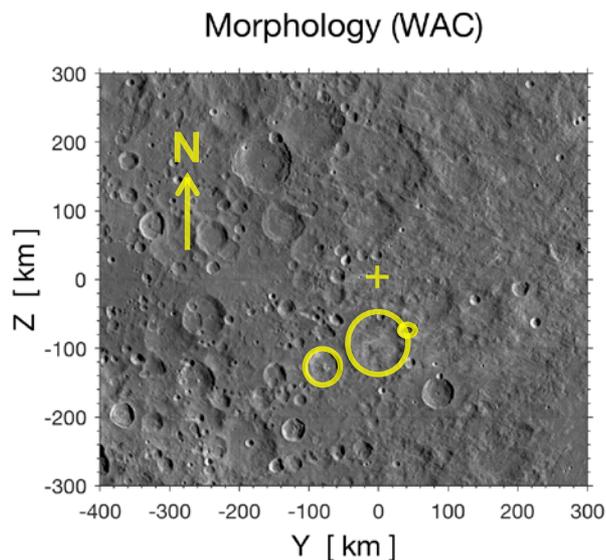
# BOLAS Science – Physical and Chemical Processes

- Oxygen
- Hydrogen



# BOLAS Science – 3D Hybrid Simulation of Gerasimovich

SURFACE



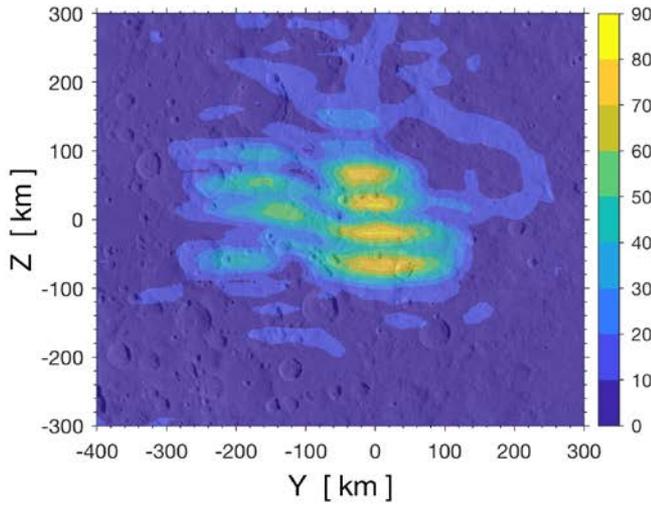
**Results from 3D self-consistent hybrid code (Fatemi et al., 2015)**

- **Uses realistic crustal field (Purucker and Nicholas, 2010)**
- **10 × 10 km resolution at the surface (Y – Z plane)**

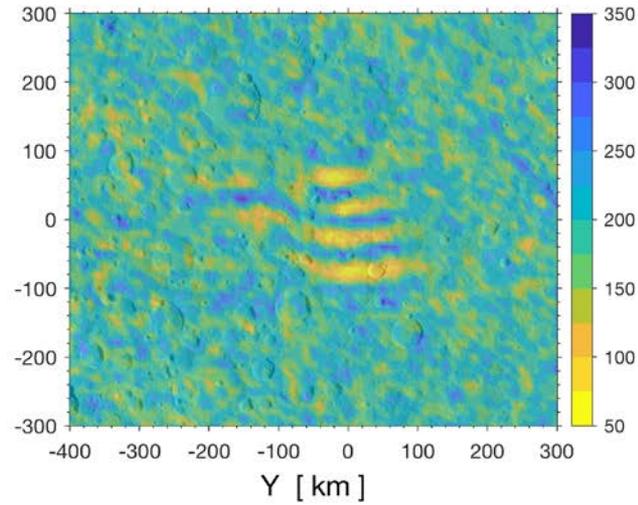
# BOLAS Science – 3D Hybrid Simulation of Gerasimovich

SURFACE

B-field magnitude [ nT ]



Down-going proton flux [  $\times 10^{10} \text{ m}^{-2} \text{ s}^{-1}$  ]

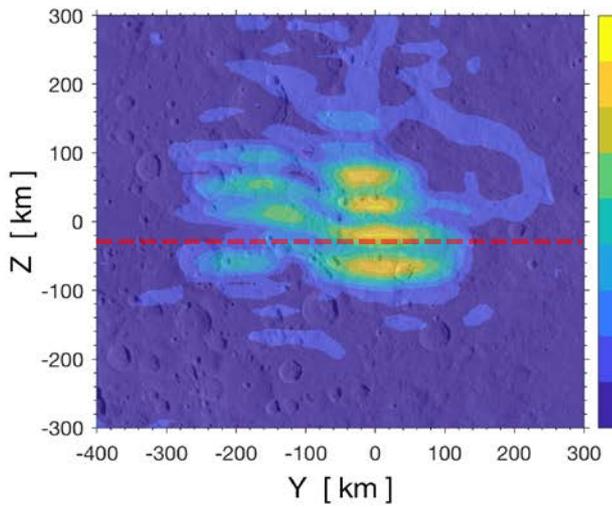


**10 km resolution  
at the surface  
captures the  
modulation of  
solar wind proton  
flux within crustal  
field region**

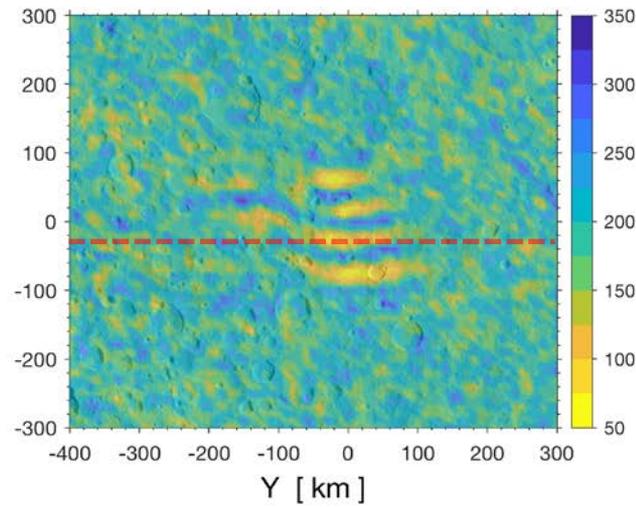
# BOLAS Science – 3D Hybrid Simulation of Gerasimovich

SURFACE

B-field magnitude [ nT ]

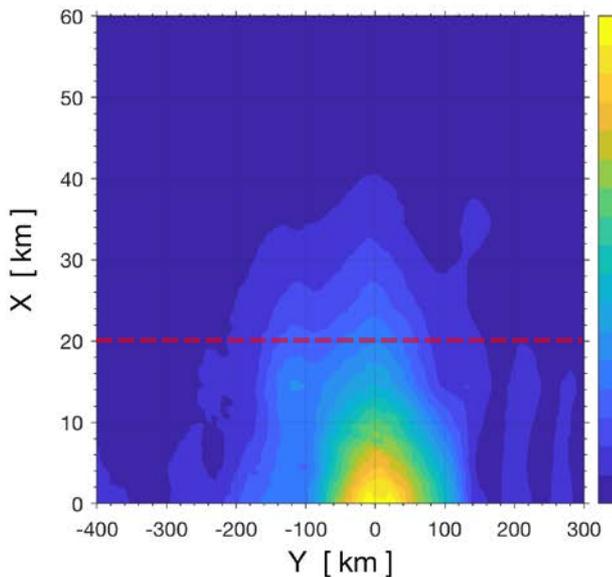


Down-going proton flux [  $\times 10^{10} \text{ m}^{-2} \text{ s}^{-1}$  ]

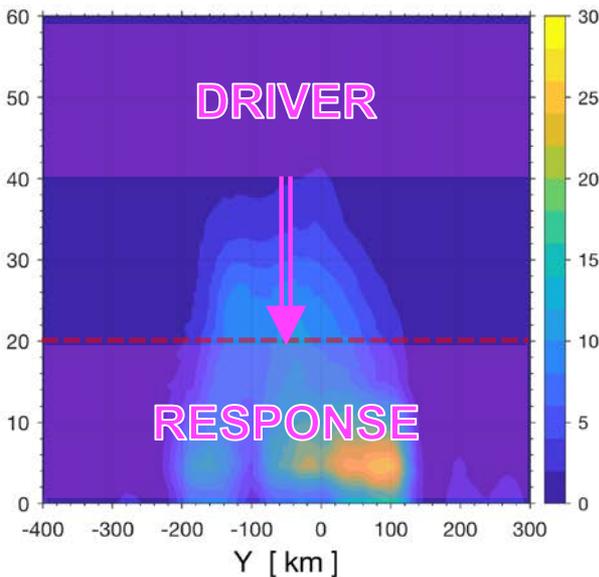


ALTITUDE PROFILE

B-field magnitude [ nT ]



E-field magnitude [  $\text{mV m}^{-1}$  ]



10 km resolution at the surface captures the modulation of solar wind proton flux within crustal field region

Altitudes  $< 20$  km required to observe processes responsible for decelerating, deflecting and reflecting solar wind protons

Vertically-aligned, dual-point measurements required to determine cause and effect

# BOLAS Instrumentation

Leveraging development of miniaturized instrumentation for CubeSats with high TRL

## Primary Payload

### **Ion Spectrometer** (Univ. of Iowa)

Incident and reflected proton energies and fluxes

### **Energetic Neutral Atom (ENA) Imager** (GSFC)

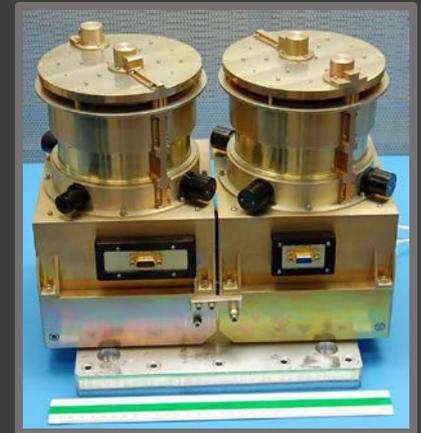
Backscattered neutral hydrogen (and ambient electrons)

### **Mini-magnetometer** (GSFC)

Ambient magnetic fields

### **Plasma Wave System** (GSFC)

Electron densities, electric fields and dust impacts



## Secondary Payload

### **Miniaturized Cameras**

Tether diagnostics & surface imaging.



# BOLAS – Frozen Science Orbit

Low-altitude lunar orbits are made difficult by “mascons” – fuel mass required for station-keeping is typically prohibitive.

“Frozen” orbits are stable, elliptical orbits that can reach low altitudes (e.g., LRO) and do not require station-keeping.

## Stable for > 1 year

- Observe annual cycles and occasional events; e.g., meteoroid streams and coronal mass ejections (CMEs)
- Full local time coverage of surface targets every 6 months

**Frozen orbit discovered with 30° inclination – equatorial – and mean altitude of 90 km**

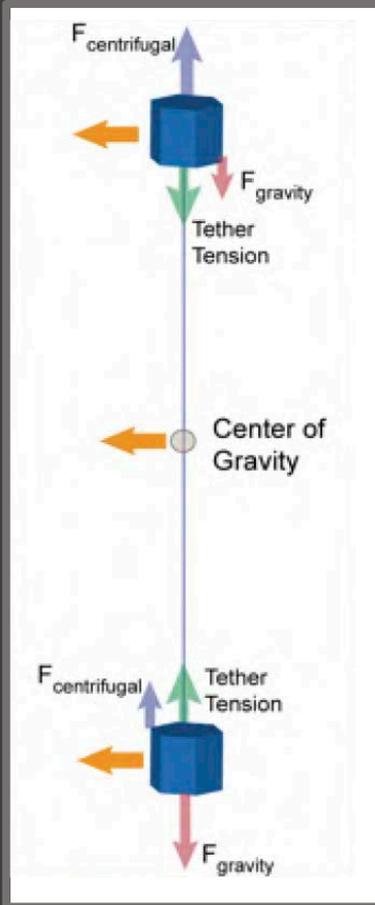
- Periapsis fixed around 30° S
- Covers all strong crustal fields ( $\geq 300$  nT at surface)
- Variable periapsis – minimum true altitude = 14.5 km

**Periapsis at Gerasimovich – minimum true altitude = 23.8 km**

**Frozen orbit not low enough over Gerasimovich for science requirements (altitude  $\lesssim 20$  km)**

# BOLAS Tethered Microsat Dynamics

Gravity gradient aligns and stabilizes tethered formation



Collier et al. (2016)

Tethered formation of two microsats enables lower spacecraft to obtain measurements closer to the Moon, while the center-of-mass of the formation follows the stable, frozen orbit.

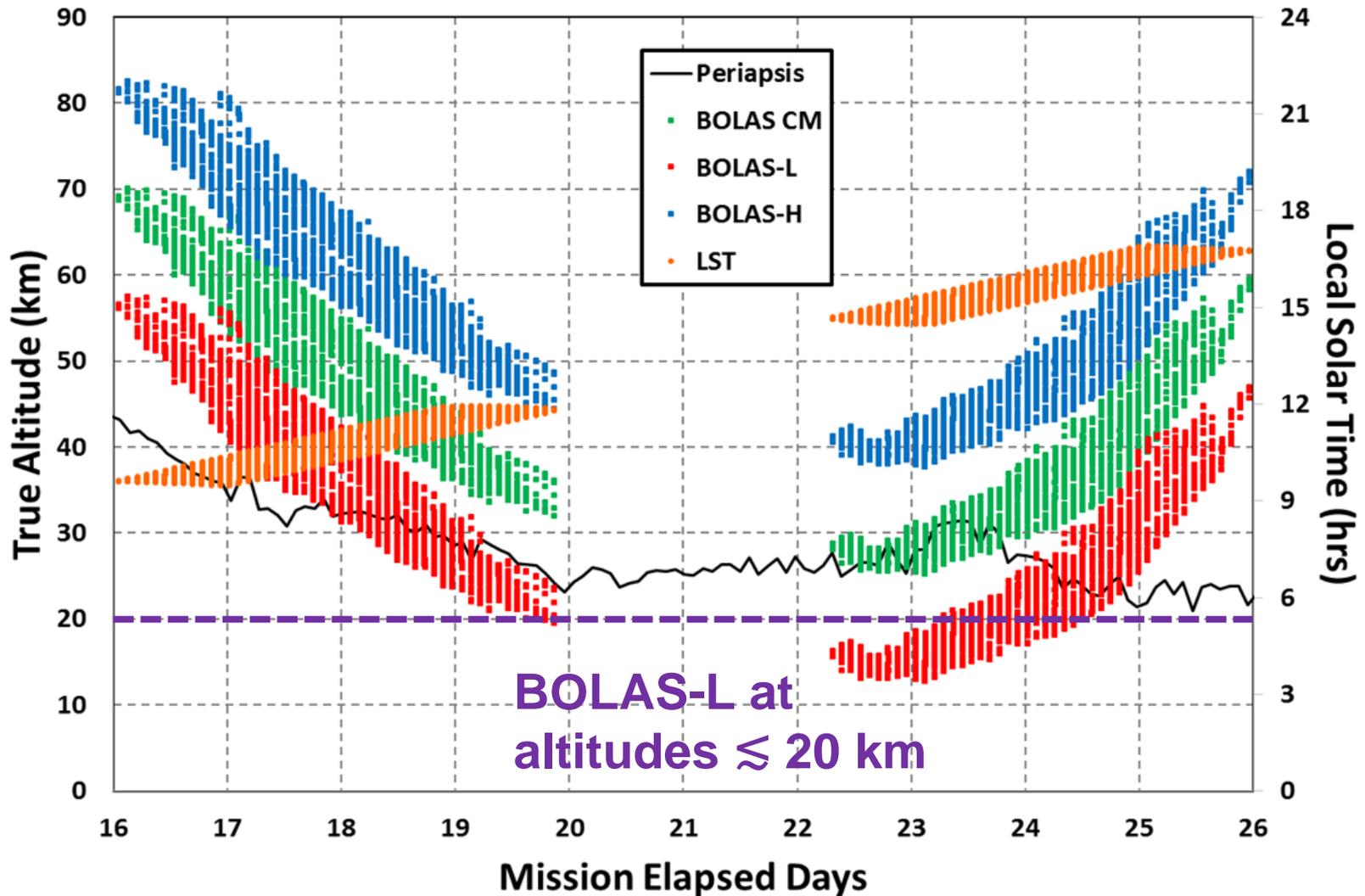
BOLAS has **25 km-long tether**, giving BOLAS-L

- Periapsis true altitude = 2 km
- Gerasimovich periapsis true altitude = 11.3 km

**Addresses BOLAS science requirements for vertically-aligned, dual-point, low altitude ( $\lesssim 20$  km) measurements at Gerasimovich**

# BOLAS – Tethered Frozen Science Orbit

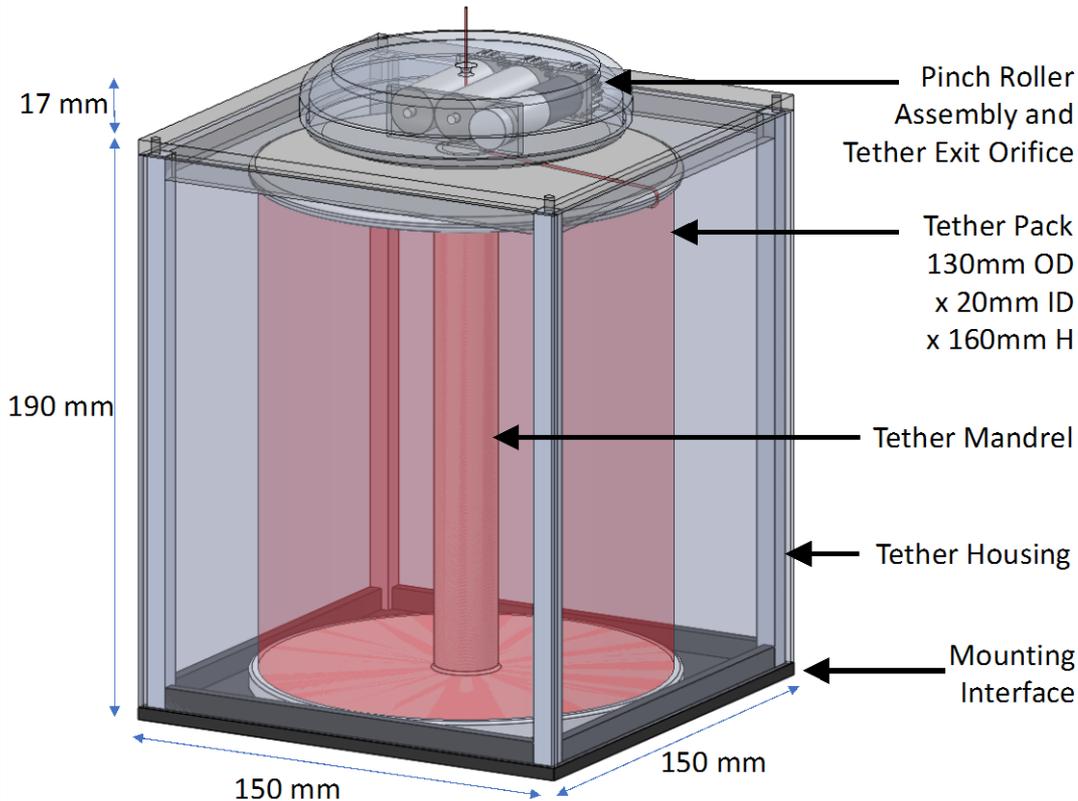
## BOLAS-Gerasimovich Overflight (90 km Frozen Orbit)



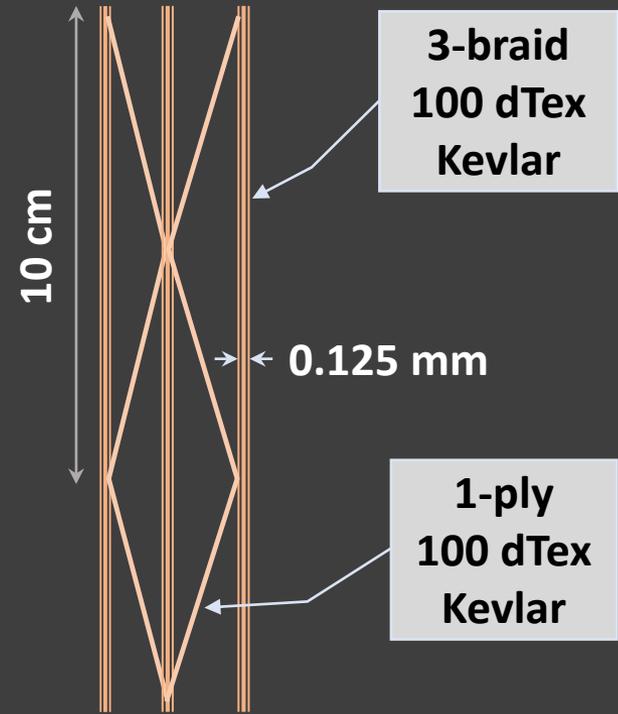
# BOLAS – Tether and Deployer

**Tether** – Kevlar yarns braided into multi-line “Hoytether” structure to provide redundancy to survive meteoroid and exospheric dust impacts

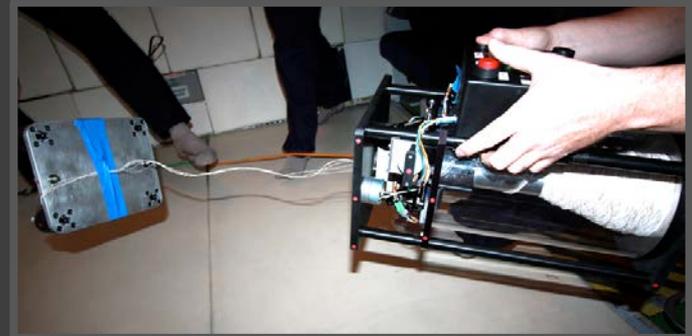
## Tether Deployer System



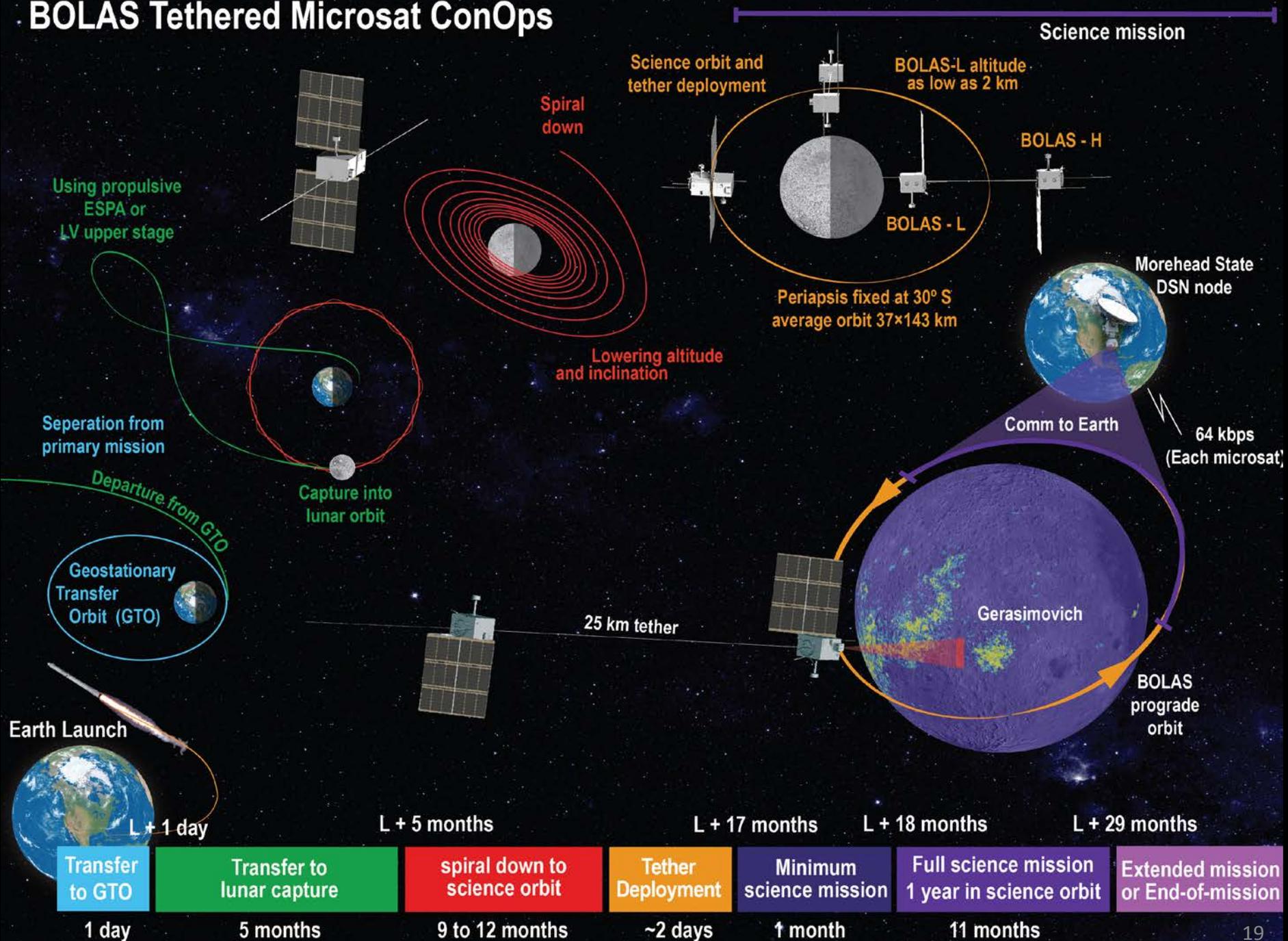
**Tethers Unlimited, Inc.**



**Similar Deployer Tested in Microgravity for LOKI Program**



# BOLAS Tethered Microsat ConOps

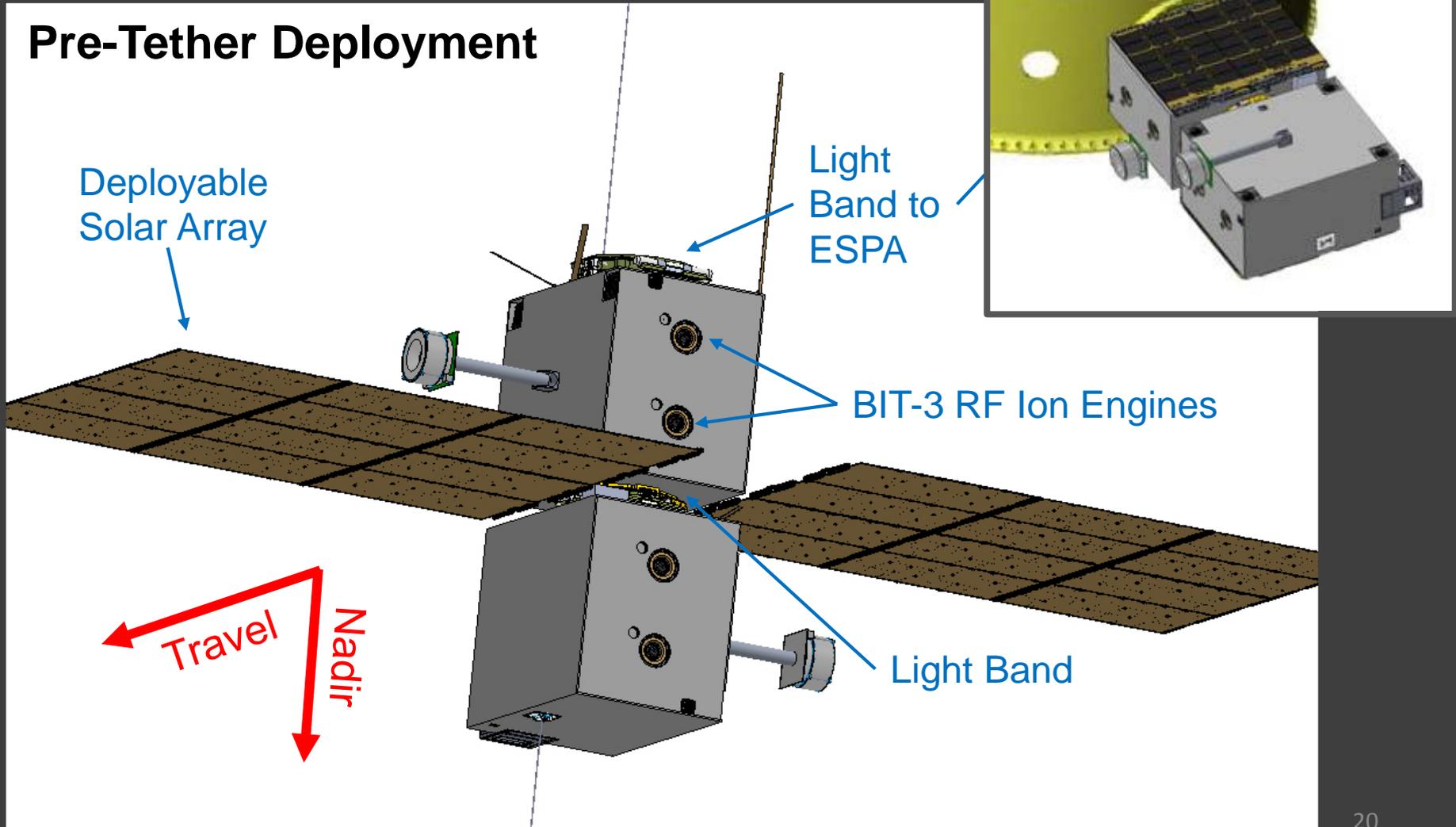


# BOLAS Microsat Design

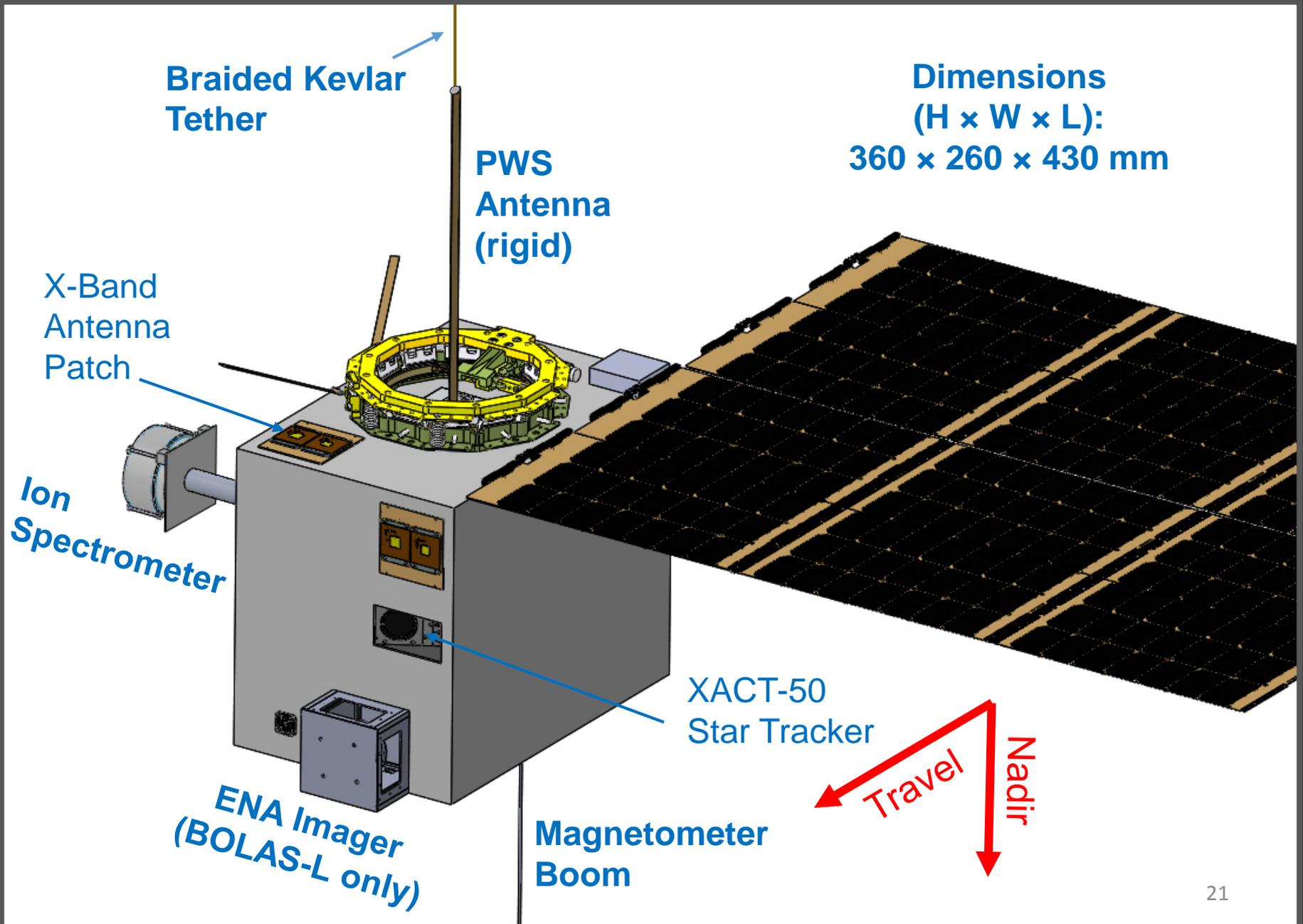
Stowed on ESPA ring

## Configuration during Spiral Down

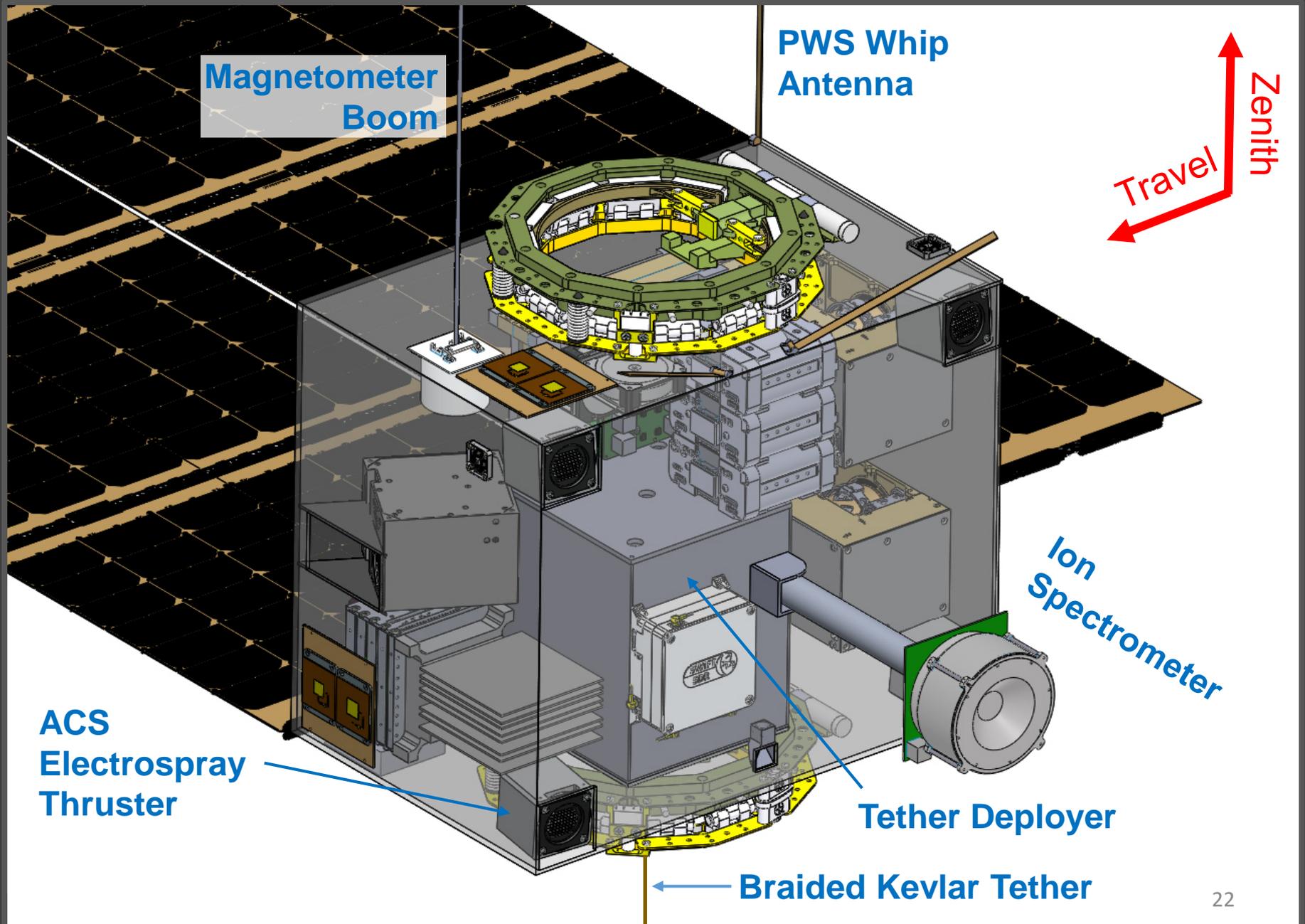
### Pre-Tether Deployment



# BOLAS-L – Post-Tether Deployment



# BOLAS-H –Post-Tether Deployment



# BOLAS Resources

Microsat	BOLAS-L	BOLAS-H
Mass [ kg ]	26.7	30.4
Peak power available [ W ]	240	240
Peak power required [ W ]	155	155
Telemetry available [ kbps ]	64	64
Telemetry required for primary science * [ kbps ]	39	39

\* Assuming: 25% transponder duty cycle (~6 h per day)  
10 / 90% split between burst and reduced modes

- **Well within ESPA-class mass and volume limits**
- **Meets power and telemetry requirements**
- **Orbit-tether-instruments exceed science requirements**

# BOLAS Tethered Microsat ConOps

